

# DEVELOPMENT OF CONTACTLESS RESONANT MEMS FORCE SENSORS IN SOI TECHNOLOGY

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**Abstract:** In this paper an innovative micromachined resonant force sensor, based on magnetic remote readout and actuation strategy is presented. The device resonance frequency changes in presence of the compressive axial-force to be measured due to changes in the device elastic properties; the sensor is driven to resonance by using externally induced Lorentz forces and the output is then obtained via an inductive pick-up.

**Keywords:** resonant force sensor, contactless sensor, MEMS.

## THE SOI FORCE SENSOR

Load cells and force sensors are generally based on bonded-foil strain gauges or piezo-resistive materials whose resistance changes when an external force is applied. These devices cannot be used in harsh environments and represent an incompatible solution for contactless and not-wired measurement systems. The microsensor presented in this work represents an innovative passive force sensor that can be interrogated adopting a remote sensing strategy: the actuation principle of the architecture proposed is based on the interaction between the eddy currents and the magnetic field generated by an external inductor driven by a sinusoidal signal; the system response (correlated to compressive axial-force) is analyzed adopting a remotely sensing system based on an inductive sensor [1].

## THE STRESS SENSOR MODEL

In the following the procedure to obtain the resonant frequency changes, as a function of the applied axial load force ( $S$ ), is briefly described using the elastic beam theory (Figure 1).

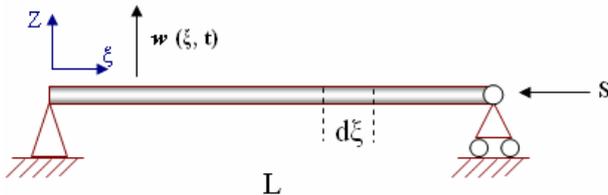


Figure 1. Beam under axial load force.

According to Figure 2 the balance equation along the z-axis, if the higher order terms are neglected, is given by eq. (1):

$$-T - md\xi \frac{\partial^2 w}{\partial t^2} + T + dT - S \sin \varphi_1 + S \sin \varphi_2 = 0 \quad (1)$$

where the  $\varphi_1$ ,  $\varphi_2$  are the angles at the initial position  $\xi$  and  $\xi+d\xi$ , respectively,  $T$  is the shear force and  $M$  is the moment.

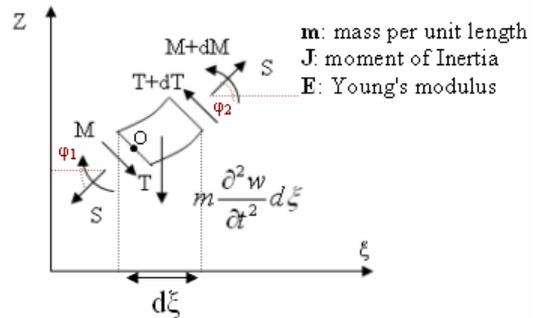


Figure 2. Force analysis.

The linearization at a static equilibrium point allows to define the left and right displacements, respectively:

$$w_L(\xi, t) = w(\xi, t)$$

$$w_R(\xi, t) = w(\xi, t) + dw(\xi, t) = w(\xi, t) + \left( \frac{\partial w(\xi, t)}{\partial \xi} \right) d\xi \quad (2)$$

where  $dw$  represents the incremental displacement as shown in Figure 3.

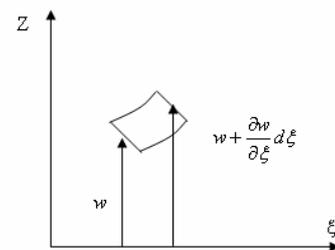


Figure 3. Incremental displacement analysis.

For small displacement ( $\varphi_1$  and  $\varphi_2$  are very small), it is possible to write:

$$\sin \varphi_1 \approx \varphi_1 \approx tg \varphi_1 = \frac{\partial w}{\partial \xi} \quad (3)$$

$$\sin \varphi_2 \approx \varphi_2 \approx tg \varphi_2 = \frac{\partial}{\partial \xi} \left( w + \frac{\partial w}{\partial \xi} d\xi \right) = \frac{\partial w}{\partial \xi} + \frac{\partial^2 w}{\partial \xi^2} d\xi$$

that replaced in the expression (1) gives:

$$\frac{\partial T}{\partial \xi} d\xi - m \frac{\partial^2 w}{\partial t^2} d\xi + S \frac{\partial^2 w}{\partial \xi^2} d\xi = 0 \quad (4)$$

The dynamic rotation around the point O can be written as follows:

$$T(\xi, t) = - \left( \frac{\partial M(\xi, t)}{\partial \xi} \right) \Rightarrow \frac{\partial T(\xi, t)}{\partial \xi} = - \frac{\partial^2 M(\xi, t)}{\partial \xi^2} \quad (5)$$

Replacing the expression (5) in the expression (4), the following equation can be obtained:

$$- \left( \frac{\partial^2 M(\xi, t)}{\partial \xi^2} \right) d\xi - m \frac{\partial^2 w(\xi, t)}{\partial t^2} d\xi + S \frac{\partial^2 w(\xi, t)}{\partial \xi^2} d\xi = 0 \quad (6)$$

Taking into consideration the Eulero-Bernoulli beam theory and under the hypothesis of homogeneous beam ( $EJ = \text{constant}$ ), a fourth-order differential equation can be obtained to describe the motion beam ( $w$ ) subjected to a static axial load  $S$ :

$$- EJ \left( \frac{\partial^4 w}{\partial \xi^4} \right) + S \left( \frac{\partial^2 w}{\partial \xi^2} \right) = m \frac{\partial^2 w}{\partial t^2} \quad (7)$$

Assuming a particular integral of the form:

$$w(\xi, t) = \psi_c(\xi) G(t) \quad (8)$$

that replaced in expression (7) gives the following differential equation:

$$EJ \frac{d^4 \psi_c(\xi)}{d\xi^4} G(t) - S \frac{d^2 \psi_c(\xi)}{d\xi^2} G(t) + m \psi_c(\xi) \frac{d^2 G(t)}{dt^2} = 0 \quad (9)$$

Then, it is possible to define two different ordinary differential equations, to obtain  $G(t)$  and  $\psi_c(\xi)$ , respectively:

$$w(\xi, t) = \psi_c(\xi) G(t) = (F_1 \sin \gamma_1 \xi + F_2 \cos \gamma_1 \xi + F_3 \sinh \gamma_2 \xi + F_4 \cosh \gamma_2 \xi) (A \sin \omega t + B \cos \omega t) \quad (10)$$

with:

$$\gamma_1 = \sqrt{-\frac{S - \sqrt{S^2 + 4EJm\omega^2}}{2EJ}}$$

$$\gamma_2 = \sqrt{\frac{S + \sqrt{S^2 + 4EJm\omega^2}}{2EJ}} \quad (11)$$

whereas  $F_1, F_2, F_3, F_4$  are strictly related to the boundary conditions and  $A, B$  correlated to the initial condition of the system at  $t=0$ ; thus, taking into account the Figure 1, they can be expressed as:

$$[w(\xi, t)]_{\xi=0} = 0$$

$$[w(\xi, t)]_{\xi=L} = 0$$

$$[M]_{\xi=0} = EJ \left[ \frac{\partial^2 w}{\partial \xi^2} \right]_{\xi=0} = 0 \Rightarrow F_2, F_3, F_4 = 0; \quad F_1 = 1$$

$$[M]_{\xi=L} = EJ \left[ \frac{\partial^2 w}{\partial \xi^2} \right]_{\xi=L} = 0$$

$$\Rightarrow w(\xi, t) = (\sin \gamma_1 \xi) (A \sin \omega t + B \cos \omega t) = \left( \sin \frac{\pi}{L} \xi \right) (A \sin \omega t + B \cos \omega t) \quad (12)$$

Equation (12) represents the beam motion model subjected to an axial load  $S$ . The frequency response as function of the applied axial load, which is represented in Figure 4, can be expressed adopting equations (11) [2]:

$$\gamma_1^2 = \frac{\pi^2}{L^2} = -\frac{S - \sqrt{S^2 + 4EJm\omega^2}}{2EJ} \quad (13)$$

$$\Rightarrow f_r \cong \frac{1}{2L} \sqrt{\frac{\left( S + EJ \left( \frac{\pi^2}{L^2} \right) \right)}{m}} \quad (14)$$

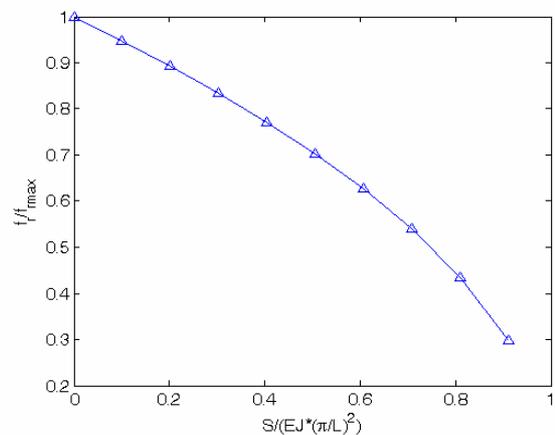


Figure 4. Frequency response as a function of the applied axial load ( $S$ ).

## THE SOI DESIGN PROTOTYPE

The force sensor has been developed by adopting a BESOI (Bulk and Etch Silicon on Insulator) technology, thus releasing a symmetric device (with aluminium surface) supported by four beams (Figure 5(a)). In order to simulate the physical behaviour of the micromachined device, CoventorWare<sup>®</sup> MemMech tool, based on a FEM method has been used (Figure 5(b) and 5(c)).

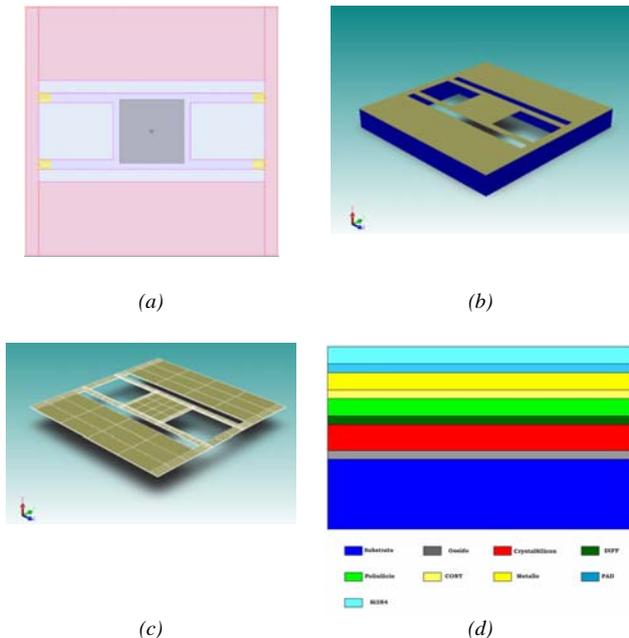


Figure 5. a) Force sensor layout in SOI technology, b) 3D model device, c) CoventorWare<sup>®</sup> mesh model, d) Silicon on Insulator stack materials.

A custom MEMS process has been adopted: SOI-based bulk micromachining (Figure 5(d)). This technology is based on a 450  $\mu\text{m}$  thick carrier silicon substrate, separated from a 15  $\mu\text{m}$  thick c-Si layer by a 2  $\mu\text{m}$  thick buried oxide.

The etching operations of Silicon from both the front side and the bottom side can be summarized as follow:

- a RIE etching on the front-side is used to remove 15  $\mu\text{m}$  of Silicon;
- a DRIE etching of Silicon on the back-side is adopted, in order to remove the 450  $\mu\text{m}$  of silicon;
- a RIE etching of buried  $\text{SiO}_2$  from the backside is used to remove the 2  $\mu\text{m}$  of oxide.

Table 1 summarizes the process characteristics.

Substrate (silicon)	450 $\mu\text{m}$
Oxide	2 $\mu\text{m}$
Silicon	15 $\mu\text{m}$
Oxide	0.1 $\mu\text{m}$
Polysilicon	0.48 $\mu\text{m}$
Oxide	0.73 $\mu\text{m}$
Metal	1 $\mu\text{m}$
Oxide	0.5 $\mu\text{m}$
Nitride	2 $\mu\text{m}$

Table 1. Feature of the custom SOI foundry process.

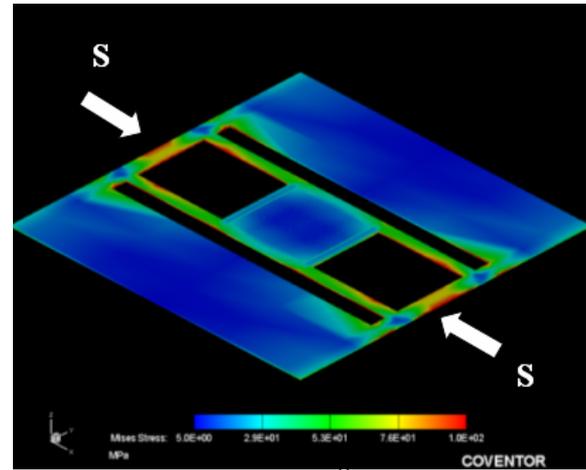


Figure 6. CoventorWare<sup>®</sup> - FEM analysis. The colour map shows the stress distribution.

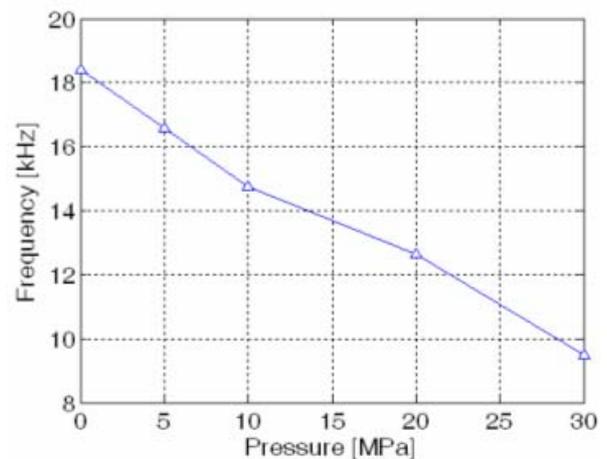


Figure 7. CoventorWare<sup>®</sup> FEM-analysis: resonant frequency as function of the axial load pressure.

## THE CONTACTLESS STRATEGY

The contactless working principle is based on Lorentz force actuation system, and a differential dual-coil inductor to measure the device resonance oscillations [3, 4]. Moreover, the interaction between the eddy currents and the magnetic field, generated by an external inductor onto the conductive mass (exerted by a sinusoidal signal) are exploited for the actuation. Furthermore, the system response (strongly depending on the applied external force) is analyzed adopting a remotely sensing based on an inductive sensor. The actuation/sensing process can be summarized as follows:

- 1) a sinusoidal bias signal, having a frequency of  $f_r/2$  ( $f_r$ : resonance frequency of the mechanical structure), is used to excite the elastic structure;
- 2) a sinusoidal signal at 80 kHz is used to bias the primary coil of the sensing inductor;
- 3) the conditioning circuit output represents the displacement information (AM signal, modulated by the device motion);
- 4) the device frequency response (related to the axial force) can be analyzed using an amplitude demodulator.

In Figure 8 the contactless readout strategy is sketched:

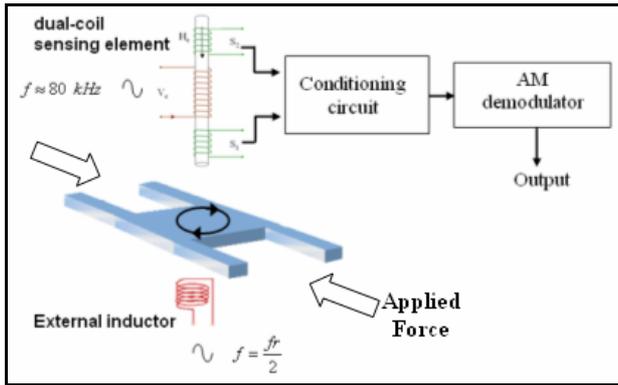
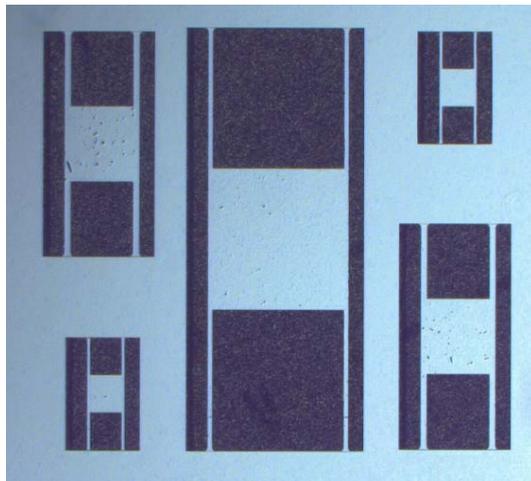
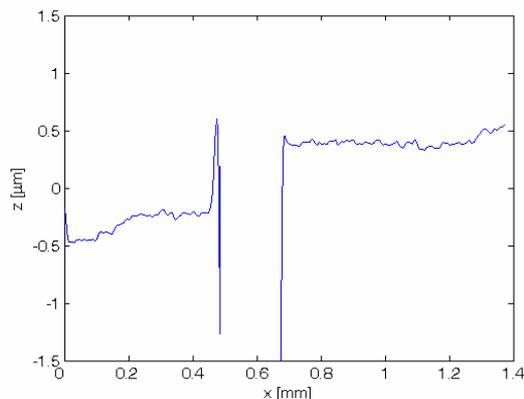


Figure 8. Contactless electromagnetic readout strategy.

The Figure 9(a) shows a microscope picture of a set of SOI force sensors while Figure 9(b) the output of a profile experimental analysis (performed through a confocal microscope).



(a)



(b)

Figure 9. a) SOI force sensor prototypes, b) profile analysis of the central pictured device.

## CONCLUSIONS

The development of a novel force sensor for contactless measurements has been described in this work.

A cantilever beam subjected to an axial load has been modeled and the frequency response, as a function of the applied axial load, has been obtained.

The developed force sensor (mainly composed by a central square area supported by four beams) has been simulated using the CoventorWare<sup>®</sup> MemMech solver, based on a FEM method.

The SOI prototypes have been realized adopting a custom MEMS process.

Future activities will include the contactless characterization of the devices, performed by using a Lorentz force actuation principle, and a sensing inductor to measure the system resonance frequency variation as consequence of the external applied axial load.

## REFERENCES

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